# PILOTS OPERATING MANUAL

# MURPHY SR 3500 MOOSE (LYCOMING)

This manual is intended as a guide only. Amateur built aircraft vary a great deal in their construction standard, equipment and handling. If you intend to fly any aircraft with which you are unfamiliar you must undertake a thorough checkout with a suitably qualified pilot.

This manual refers specifically to the Murphy SR3500 (MOOSE) with a maximum gross weight of 3500 lbs.

Fixed tailwheel landing gear and powered by the Lycoming 540 series 6 cylinder engine.

All information is given assuming a factory standard Super Rebel with no modifications.

<b>KIT</b>	No.		
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#### **SECTION 1**

#### **LIMITATIONS**

Maximum gross weight 3500 lbs. This aircraft is not approved for aerobatics or spins.

#### **ENGINE**

LYCOMING 540 series 6 cyl. 250 to 300 H.P. Refer to Lycoming Operators Manual.

#### AIRSPEEDS FOR EMERGENCY OPERATION

Engine failure after takeoff	75 mph75 mph.
OPERATIONAL CHECKLIST	
Engine Failure During Takeoff a) Throttle b) Brakes c) Flaps d) Mixture e) Mags f) Master switch g) Door	applyretractidle cut offoff.

Engine Failure Right After Takeoff a) Airspeed b) Mixture c) Fuel valves d) Mags e) Flaps f) Master g) Door	idle cut offoffoffas requiredoff.
Engine Failure During Flight  a) Airspeed, trim to b) Carb heat, alt. air c) Primer (carb engine) d) Boost pump e) Electrics f) Left fuel g) Right fuel h) Mixture i) Restart	on. in & locked. on. on. on. on. on. rich.
FORCED LANDING Emergency Landing Without Engine Power	
a) Trim b) Mixture c) Fuel valves d) Mags e) Flaps f) Master g) Doors	flaps dn 65 mphoffoffoffoffoffoff.

#### **FIRE**

# **Engine Fire On Start**

- a) Continue cranking to suck fire & fuel into engine. If engine starts, run at 1800 for 1 min. then shut fuel off.
- b) Engine.....shut down inspect for damage.
- c) If engine fails to start continue cranking while shutting off fuel, Mixture, mags and master.
- d) Obtain fire extinguisher.
- d) Evacuate aircraft.
- e) Extinguish fire.

### **Engine Fire In Flight**

a) Fuel	off.
b) Cabin air/heat	close.
WHEN ENGINE STOPS	
c) Throttle	close.

- d) Mags......off.
- e) Mixture......off.
- f) Airspeed.....increase to find incombustable mixture.
- g) Forced landing.....execute.

# Electrical Fire In Flight

- a) Master......off.
- b) Cabin air/heat......close.
- c) Extinguisher......activate.
- d) Ventilate Cabin

#### AIRSPEED LIMITATIONS

Arc colour	KTS.	M.P.H.	Operating range
White arc	46-75	53-86	Full flap operating range
Green arc	52-130	60-150	Normal operating range
Yellow arc	130- 164	150- 189	Operate with caution, only in smooth air
Red line	164	189	NEVER EXCEED

Speed	KTS	M.P.H.	Remarks
Never exceed	164	189	Do not exceed
V <sub>ne</sub>			
Max. structural			Do not exceed
cruising V <sub>no</sub> or	130	150	except in
V <sub>c</sub>			smooth air.
Manoeuvring	109	125	no full or abrupt
V <sub>a</sub>			control
			movements
			above this
			speed.
Max. flap ext.	75	86	No full flaps
$V_f/V_{fe}$			above this
			speed.

# CENTRE OF GRAVITY CONVENTIONAL GEAR

The allowable center of gravity ranges for all Super Rebels and Moose models is from 10.8" to 20.4" aft of the datum line.

Forward limit = 10.8" A.O.D. Aft limit = 20.4" A.O.D. To determine empty weight center of gravity:

- 1/ Place aircraft in level flight attitude, with a suitable scale under each wheel.
- 2/ Record weight on each wheel.
- 3/ Measure the distance from main wheel center to tail wheel center, 90° to main axle.

## Sample Empty Weight:

Scale position	Wt.lbs.	Arm,inches	Moment
Left main	810	- 1.5"	-1215
Right main	806	- 1.5"	-1209
Tail wheel	74	242"	17908
Totals	1690	9.16"	15484

15484 divided by 1690 = 9.16 inches aft of datum.
THIS IS THE EMPTY WEIGHT CENTER OF GRAVITY

# Forward sample loading:

	Wt.lbs	Arm,inches	Moment
Empty Weight	1690	9.16"	15484
Front Seat, pilot	176	15"	2640
Rear seat	0	50"	0
Fuel, 20 gal.	120	29"	3480
Baggage	0	70"	0
Totals	1986	10.88"	21604

Total moment 21604 divided by total weight 1986 = 10.88" aft of datum. This is between the limits of 10.8" and 20.4". The aircraft is at an acceptable weight and balance for flight.

## Rear sample loading:

	Wt. Lbs	Arm, inches	Moment
Empty weight	1690	9.16"	15484
Front seats	400	15"	6000
Rear seats	300	50"	15000
Fuel	480	29"	13920
Baggage	130	70"	9100
Totals	3000	19.83"	59504

Total moment 59504 divided by total weight 3000 = 19.38" aft of datum. This is between the limits of 10.8" and 20.4". The aircraft is at an acceptable weight and balance for flight.

#### **GENERAL SPECIFICATIONS**

GROSS WEIGHT. EMPTY WEIGHT. USEFUL LOAD. G – LOADING (ultimate). FUEL CAPACITY. WING SPAN. WING AREA. WING LOADING. LENGTH. FUSELAGE WIDTH. WIND CORD. TAIL SPAN. TAIL CORD. AIRFOIL	
AIRFOIL	modified 4415
FLAP AREA	142740.
Simple Flaps	

PERFORMANCE SPECIFICATIONS AT GRO	DSS WEIGHT (3000 lbs)
NOTE: Based on standard temperature 15 C	at sea level
ENGINE, Lycoming IO-540-D4A5	260 H.P.
POWER LOADING	11.54 lbs.hp
TAKE OFF RUN	600 ft.
LANDING ROLL	
RATE OF CLIMB	1200 ft. min.
FULL FLAP STALL, power off	53 mph.
NO FLAP STALL, power off	60 mph.
CRUISE (75% power)	147 mph.
TOP SPEED	155 mph.
FUEL BURN (75% power)	13.5 US gal/hr.
ENDURANCE	6 hrs.
RANGE	870 st mi.
FUEL CAPACITY (standard)	80 US gal.

#### NORMAL PROCEDURES

#### PREFLIGHT INSPECTION

#### COCKPIT

- a) Control locks (if fitted) removed.
- b) Magneto switch OFF keys out.
- c) Master switch ON.
- d) Check strobes, landing light, fuel guages, nav. lights, and electric trim at neutral (green).
- e) Master Switch OFF.
- f) Operate flaps through range. Leave at full flap.
- g) Fuel taps both ON.
- h) First aid kit and fire extinguisher in place.
- I) Req. documents on board.
- j) Record hobbs

#### FRONT

- a) Cowling secure, intake clear.
- b) Propeller for security and nicks.
- c) Spinner for security and cracks.
- d) Exhaust for security and cracks.
- e) Oil contents 12 qts. Long trip, min 9 qts)
- f) Inspect front sump fuel sample.
- g) Inspect windshield for cracks and security.

#### LEFT FRONT

- a) Inspect front side panel and lower lift strut area for damage and security.
- b) Inspect landing gear leg for signs of for and aft movement and security.
- c) Inspect brake lines for fraying and leakage.
- d) Inspect brake pads for wear and rotors for security.
- e) Inspect wheels and axle for security.
- f) Inspect tires for proper inflation and wear.

#### LEFT WING

- A) Inspect left wing leading edge for damage.
- B) Inspect wing tip and lights.
- C) Inspect aileron operation and security.
- D) Inspect flap operation and security.
- E) Inspect upper lift strut for security.
- F) Take and inspect fuel sample from left wing.
- G) Inspect fuel quantity, vent and cap security.
- H) Pitot inlet clear.

#### LEFT SIDE

- a) Inspect door operation and latch for security.
- b) Inspect window for operation and security.
- c) Baggage securly tied down.
- d) Baggage door secure and locked.
- e) Inspect antennas for security.

#### TAIL GROUP

- a) Fin and fin fairing secure.
- b) Horizontal stabilizer secure.
- c) Elevator secure.
- d) Elevator hinge bolts and pins secure.
- e) Elevator trim tab secure.
- f) Rudder hinge bolts and pins secure.
- g) Rudder trim secure if applicable.
- h) Rudder horn for freedom of movement and security.
- i) Rudder cables and turnbuckles for safety and security.

#### TAILWHEEL

- a) Tailwheel stinger for damage and security.
- b) Tailwheel for security and condition.
- c) Tailwheel springs and chains for wear.
- d) Tailwheel swivel for freedom of movement.

#### RIGHT SIDE

- a) Inspect right side of fuselage.
- b) Inspect window and door for operation and security.
- c) Take and inspect fuel sample from fuselage bottom.
- d) Inspect antennas for security.

#### **RIGHT WING**

- a) Take and inspect fuel sample from right wing.
- b) Inspect flap for security and operation.
- c) Inspect aileron for security and operation.
- d) Inspect wing tip and lights.
- e) Inspect leading edge for damage.
- f) Inspect upper lift strut for security.
- g) Check fuel quantity, vent and cap security.

#### RIGHT FRONT

- a) Inspect front side panel and lower lift strut area for security.
- b) Inspect landing gear leg for signs of for and aft movement.
- c) Inspect brake line for fraying and leakage.
- d) Inspect brake pads for wear and rotors for security.
- e) Inspect wheels and axles for security.
- f) Inspect tires for proper inflation and wear.

BEFORE START  a) Passengers
START (Lycoming O-540 series)  a) Carb heat
AFTER START  a) R.P.M

# TAXI

a) Brakes	check.
b) Steering	check.
c) Heading	check.

# RUN UP

a) Position	. Into wind & clear all round.
b) Brakes	on.
c) Doors & windows	closed.
e) Flight controls	up elevator.
f) Flight instuments	check.
g) Fuel	on.
h) Boost pump	off.
i) R.P.M	1700
Magnetos check:	
LEFT - RIGHT - BOTH	
Max. drop	
Max. diff. between mags	
j) Engine instruments	
k) Prop	
l) Carb heathotmax. drop 7	
n) R.P.M Idle 500 - 700.	Reset 1000
DDE TALKE OFF AND MITAL ACTIONS	
PRE-TAKE OFF AND VITAL ACTIONS	
a) Trim	act for take off
a) Trim	
b) Throttle	
d) Fuel	
e) Prop	
f) Flap	
g) Instruments	
h) Loose articles	_
11/ 10000 01110100	stowed.

i) Seatbeltssecure.j) Controlsfree & correct.k) Transponderon.l) Lookoutproceed when clear.
Normal Takeoff a) Flaps
Short Field Take Off  a) Flaps
Enroute Climb
a) Airspeed.       .90-100 mph.         b) Flap.       .1 notch.         c) Throttle.       .full.         d) Mixture.       rich below 3000 ft.

# LANDING CHECKS

a) Mixture
Airspeed and flaps
a) Downwindflaps 1
Crosswind technique - Wing low, use aileron to keep wing down, USE BRAKES TO KEEP STRAIGHT.
AFTER LANDING
a) Carb heat
SHUT DOWN
a) Brakes

# NOTES